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Rotorcraft's Hot Spot Monitoring Using Distributed Smart Patch System

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Utilization of structural health monitoring (SHM) system is by far the most cost effective means to provide health status of rotorcraft structures. However, there are many challenges in the pathway to implement the flight qualified SHM system. A few of them are installation of sensors, optimizing the number of sensors, ensuring reliable functionality of the diagnostic hardware, reliable detection, localization and quantification of respective damage using diagnostic software etc. Since the rotorcrafts faces extreme environmental conditions during their operation, the functionality and reliability of the SHM system needs to be ensured under variable flight conditions. Moreover, different rotorcrafts components are subjected to different geometry and loading conditions. Thus, implementation of the system is also different. Components with fatigue crack and components prone to have corrosion needs to be treated differently. However, it is of great interest to develop a unique multipurpose system for the rotorcrafts. Thus, in this paper a few examples on fatigue crack monitoring using Acellent' SmartPatch System is demonstrated. A helicopter's bifilar, a specified hot spot in a helicopter rare extension and a composite head beam were investigated in great detail. Damage detection, localization and quantification of damages are presented in this paper.

Ключевые слова:

Содержание.

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